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Determination of Stress Intensity Factor for a Plate Having Square Hole with Inclined Crack using Numerical Method

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Abstract: The objective of this work is to develop a new material model to simulate the fracture behavior of structural plate with hole. Plates are subjected to different types of loading like tension, compression, bending or any combination of these. These different types of loading situation may initiate and propagate a crack. The issue of crack detection and fracture has gained wide spread industrial interest. Crack or damage affects the industrial economic growth. Generally crack in a structural element may occur due environmental defects and during manufacturing of the plates. A finite element based two dimensional crack propagation simulator software ANSYS is used for the analysis. Finite element analysis was employed to simulate the fracture behavior of the plate with inclined crack. The stress intensity factor is an important parameter for estimating the residual life in cracked structures. To determine the mixed-mode stress intensity factors quarter-point (Q-P) singular finite elements are employed. Crack growth depends on the initial crack length, material properties and dimensions, loading conditions etc. So Load versus crack length increase and load versus potential energy are calibrated.

Key words: stress intensity factor, mixed-mode, quarter-point and singular elements.

INTRODUCTION

Aluminium, the second most plentiful metallic element on "Determination of Stress Intensity Factor for a plate earth, became an economic competitor in engineering having square hole with inclined crack using numerical applications as recently as the end of the 19th century. method". And predict the direction of crack with different Aluminium played the role of an automotive material of material properties and loading condition. increasing engineering. In recent times aluminium materials are used for aerospace applications and fracture failure been identified. Surface and corner cracks are encountered in engineering structures and in aerospace > Preparation of 2D geometric model using Ansys applications at locations where high stresses or material imperfections exist. Sometimes, surface or corner cracks > can also be observed in a component even before its service life begins.

Fracture is the separation, or fragmentation, of a solid > At constant angle and constant load, increase the crack body into two or more parts under the action of stress.

PROBLEM FORMULATION

A plate with hole is analyzed to find the stress concentrations around holes, normally causes failure and have a great practical importance. The analysis of holes are imperative because the holes are used in engineering components such as bolts, rivets etc. and the stresses and deformation which occur near them at given load are analyzed. Due to the increase in stress above the Figure 3.2 shows the geometric model. The rectangular maximum yield stress around the vicinity of the hole, crack starts to initiate, as the load crosses the threshold value the crack starts to propagate and structure will fail. To predict this crack better study is required

OBJECTIVE

- package.
- Calculating the stress, displacement, and strain energy of the geometric model by applying boundary conditions.
- length and find the stress, displacement, energy release rate and stress intensity factor.
- > By applying varying load at different angles of the crack find the stress, strain energy, displacement, energy release rate and stress intensity factor.
- \succ By using different materials, find the material behavior at different loads, crack length.

GEOMETRIC MODEL

plate of height 25mm and width 27.75mm.

The square hole inside the plate is of 6.25mm, initial crack length a_1 is 1.5625mm and crack length a_2 is 4.6097mm.

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Figure 3.2 Geometric Model

MESHING

In finite element model two lines are highlighted which is shows the crack length. Horizontal crack length denoted by a_1 and inclined crack length denote by a_2 .Inclined crack length made angle to the Y axis is 40° as shown in geometrical model.



Figure 3.3 Geometric Model



Figure 3.4 Enlarged view of crack tip

3.2.8 Two conditions are used for the analysis of the plate.

- Uniaxial load
- Biaxial load



Figure shows the geometric model of uni axial load condition



Figure shows the geometric model of biaxial load condition

RESULTS AND DISCUSSION

By varying the crack angle and with different load condition the stress and energy release rate has been determined. We have considered two materials, mild steel and aluminium in this analysis. The material properties are shown in table 4.1.

Table: 4.1 Properties of the materials
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Properties	Aluminum	Mild Steel	
Elastic Modulus	70-79 GPa	210GPa	
Poisson's Ratio	0.33	0.28	
Density	2600-2800 kg/m ³	7850 kg/m^3	

CALCULATE THE STRESS NEAR THE CRACK TIP AT VARIOUS ANGLE WITH CONSTANT CRACK AND LOAD.

4.1.1 Aluminium with uniaxial loading.

The values of crack tip stress, displacement, and strain energy for the boundary conditions shown in figure 4.1 are tabulated in the Table 4.2.



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Crack length a_1 (10 ⁻²)	Crack length a_2 (10 ⁻²)	Angle θ	Crack tip stress N/m ²	Strain energy (Ø) J	Displacement (Δ) mm
1.5625	4.6097	30	363.31	1.565	0.27779
1.5625	4.6097	35	376.38	1.564	0.27756
1.5625	4.6097	40	387.06	1.563	0.27753

Table 4.2 Variation of strain energy and displacement with varying crack angle





Figure 4.5 shows the Von-mises stress plot for 30[°] and 55[°] crack angles. It can be observed from the figure that the TIP AT VARIOUS ANGLE WITH CONSTANT maximum stress of 363.31N/m² and 401.03N/m² was found CRACK AND LOAD. at the crack tip for 30° and 55° crack angles respectively. **4.2.1 Mild steel with uniaxial loading.** At the crack tip maximum stress is shown which is By keeping the crack length and load as constant and indicated by red colour and blue colour indicates stress is increasing the crack angle for the plate with square hole, minimum or less at that region.



Figure 4.5 Von-mises stress plot for 30° crack angle.

CALCULATE THE STRESS NEAR THE CRACK

the variation in stress and strain were tabulated in table 4.3.

Crack length $a_1(10^{-2})$	Crack length $a_2(10^{-2})$	Angle 0	Crack tip stress N/m ²	Strain energy (Ø)J	Displacement (Δ) mm
1.5625	4.6097	30	374.32	5.432	0.0964
1.5625	4.6097	35	385.34	5.427	0.09632
1.5625	4.6097	40	394.15	5.425	0.00963

Table 4.3. Variation of strain energy and displacement with varying crack angle



Figure 4.8 shows the variation of crack tip stress with varying crack angle. It can be observed from the figure 4.8

that at a crack angle of 30^{0} a minimum crack tip stress of 374.32 N/m² was present and varied linearly till the crack angle of 55[°] with a maximum crack tip stress of 406.67 N/m^2 .





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the maximum stress of 374.32 N/m^2 and 406.67 N/m^2 stress is minimum or less at that region. was found at the crack tip for 30° and 55° crack angles

Above figure shows the Von-mises stress plot for 35° and respectively. At the crack tip maximum stress is shown 45[°] crack angles. It can be observed from the figure that which is indicated by red colour and blue colour indicates

STRESS INTENSITY FACTOR FOR A CRACK LENGTH UNDER VARIOUS BIAXIAL LOADS.

Table: 4.13 The stress intensity factor for varying load.

4.5.1 Aluminium with biaxial load

Load N	PotentialenergyatCracklength4.6097 (Π) J	Potential energy at Crack length 4.950 (Π) J	Changeinpotential energy(d∏)	Energy release rate (G) Jm ²	Stress intensity factor (k) MPa√m
σ_1 -4000, σ_2 -2500	4.82	4.97	-0.15	44.03	1.85
σ_1 -5000, σ_2 -3000	7.35	7.56	-0.21	61.65	2.20
σ ₁ -7500, σ ₂ -5000	17.76	18.27	-0.51	149.73	3.42

500 450 400 Crack tip Stress (N/m²) 350 300 250 200 Crack tip Stress Crack tip Stress

Load (o) N

Figure 4.27 Load v/s crack Tip Stress.

7500/5000

11000/8000

5000/3000

crack tip stress 166.80 N/m² for crack lengths of 4.6097 cm at a load of σ_1 -4000N, σ_2 -2500N and 168.04 N/m² for crack lengths of 4.9503 cm at a load of σ_1 -4000N, σ_2 -2500N and increase linearly till the crack length of 4.6097cm and 4.9503 cm at a load of σ_1 -4000N, σ_2 -2500N.

Figure 4.27 shows the comparison of the variation of the

STRESS INTENSITY FACTOR FOR A CRACK LENGTH UNDER VARIOUS BIAXIAL LOADS.

Mild steel with bi axial load condition

Figure 4.31 shows the comparison of the variation of the crack tip stress for crack lengths of 4.6097 cm and 4.9503 cm respectively.

Load N	Potential energy at Crack length 4.6097 (П) J	Potential energy at Crack length 4.6097 (П) J	Changeinpotentialenergy (d∏)	Energy release rate (G) Jm ²	Stress intensity factor (k) MPa√m
σ_x -4000, σ_2 -2500	1.79	1.84	-5	14.67	1.82
σ_1 -5000, σ_2 -3000	2.73	2.81	-8	23.48	2.31
g7500 g5000	6.62	6.80	-18	52.84	3.46

Table: 4.13 The stress intensity factor for varying load.



Figure 4.31 Load v/s crack Tip Stress

CONCLUSION

CONCLUSION BASED ON 5.1 ANALYTICAL STUDY:

Mixed-mode stress intensity factor solutions were presented for inclined cracks in finite thickness plates under uniform tensile loading conditions. As the numerical solution tool, ANSYS -11 was used, different crack lengths, crack angles and different load conditions were considered and plate was analyzed.

With the variation of load results in increases of the crack tip stress, potential energy, energy release rate and stress intensity factor.

150

4000/2500

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- The variation in strain energy with the change in crack angle was shown. It was observed that the strain energy had a decremented slope for 30⁰, 35⁰ and 40⁰ angle of crack while incremental slope for 45⁰,50⁰ and 55⁰ angle of crack.
- The variation in displacement with the change in crack angle was shown. It was observed that the strain energy had decremented slope from 30^0 , 35^0 and 40^0 angle of crack while incremental slope for 45^0 , 50^0 and 55^0 angle of crack.
- In the polar plot the stress was maximum at an angle of 120⁰ to 180⁰ and the crack will grow in the direction where the stress is maximum.
- For the biaxial load for the different material with increase in crack length the crack tip stress, strain, displacement and potential energy increase.

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